



# **Multidisciplinary Design Optimization of a Composite Amphibious Aircraft Fuselage**

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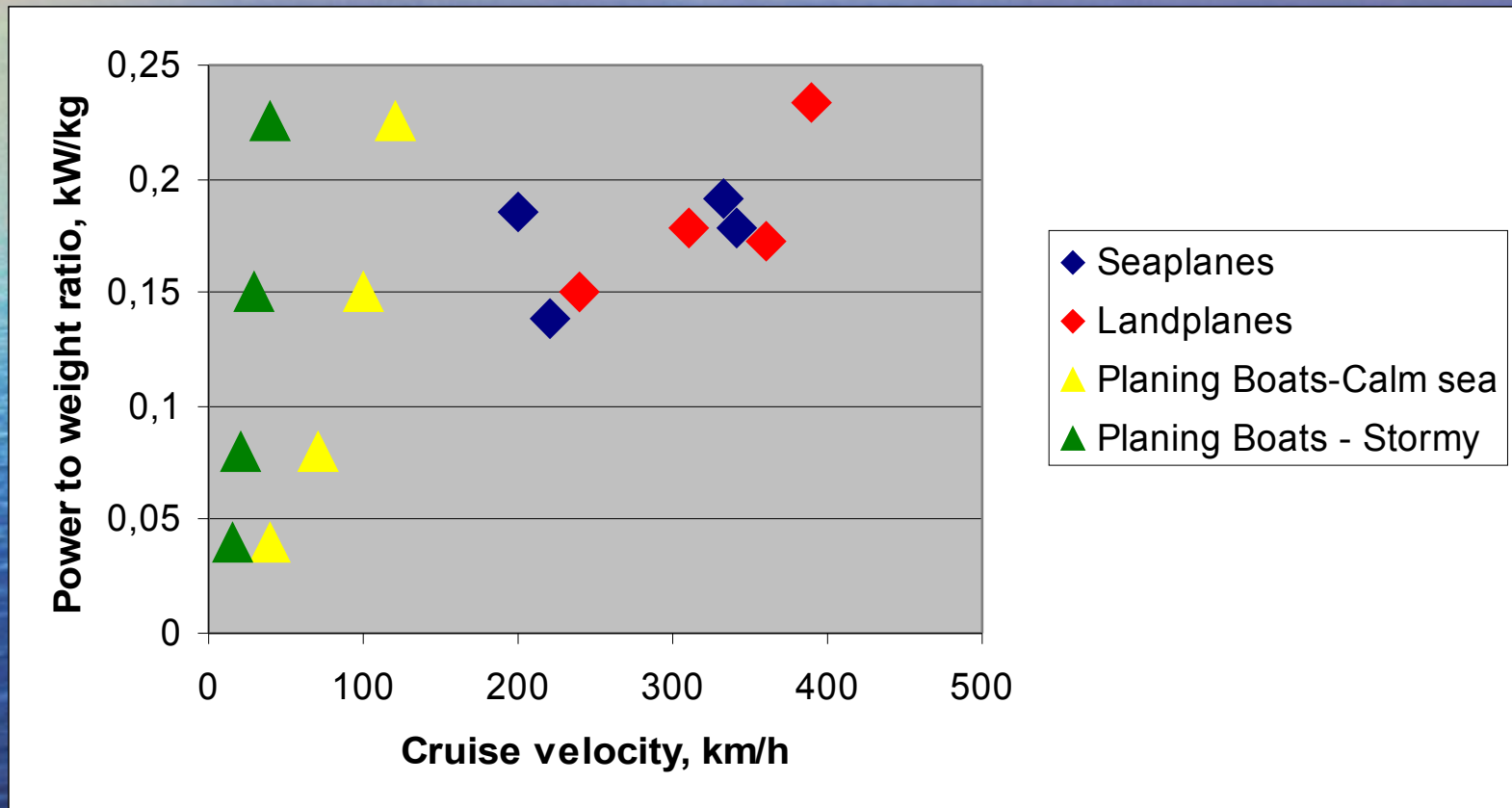
# The golden age of seaplanes is long gone...

because of:

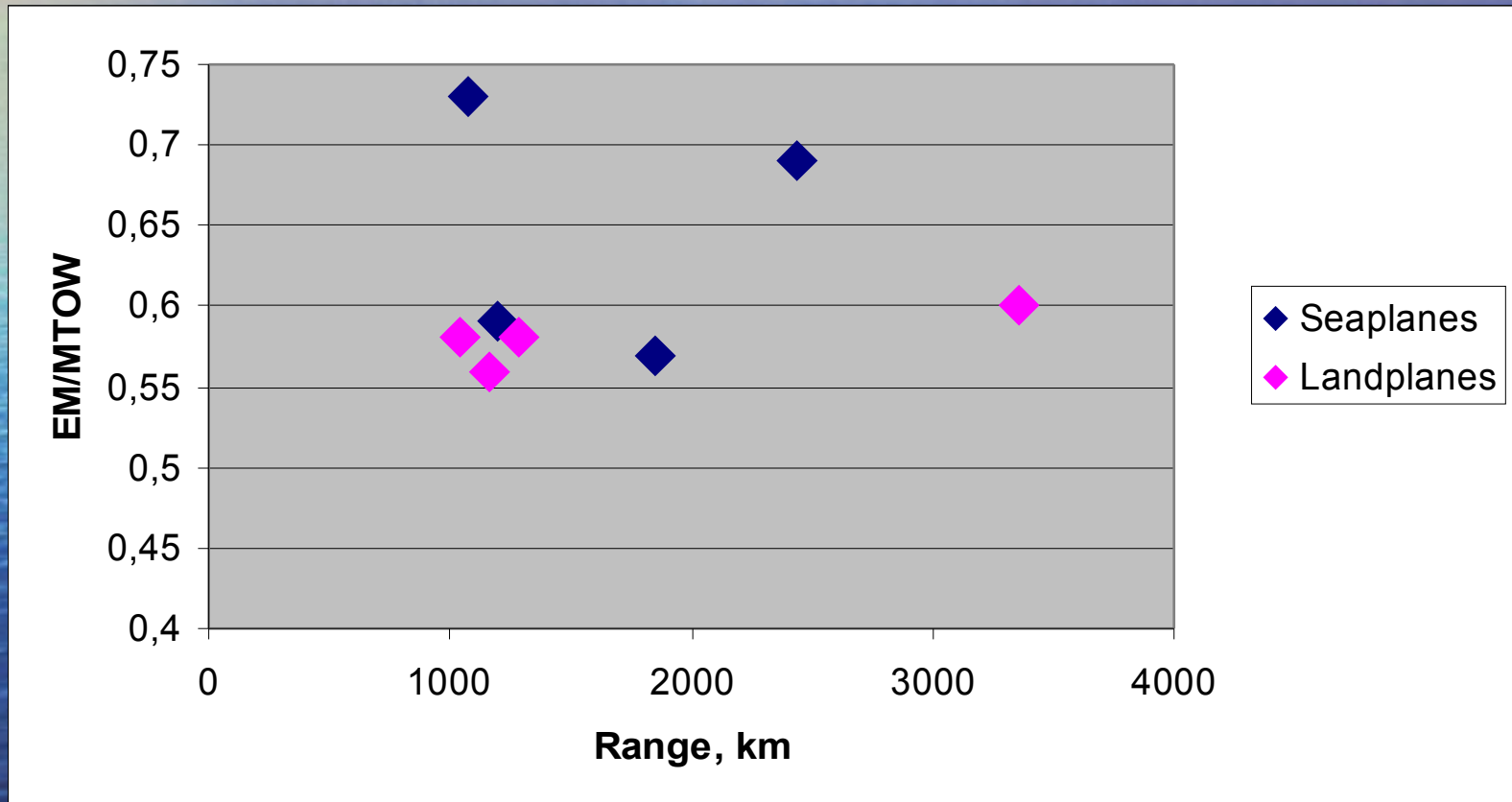
- Higher weight
- Higher drag
- Corrosion



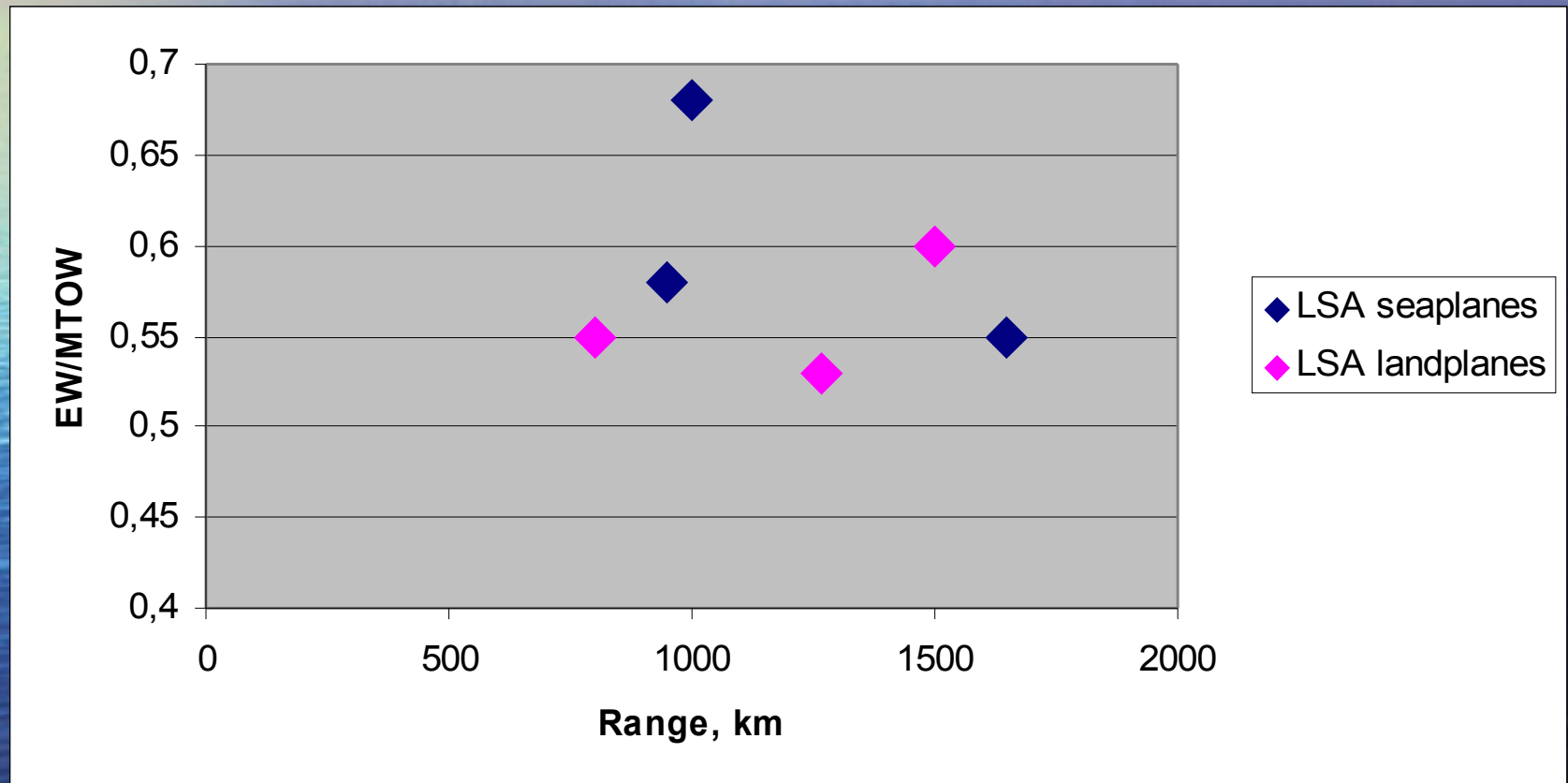
# Power-to-weight ratio of planing boats and airplanes



# Empty weight to Maximum take-off weight of commercial seaplanes and landplanes



# Empty weight to Maximum take-off weight of LSA seaplanes and landplanes



# Advantages of composite structures for amphibious aircraft

- Eliminate corrosion
- Reduce weight
- Cheaper maintenance and longer life
- Improved shape – lower drag

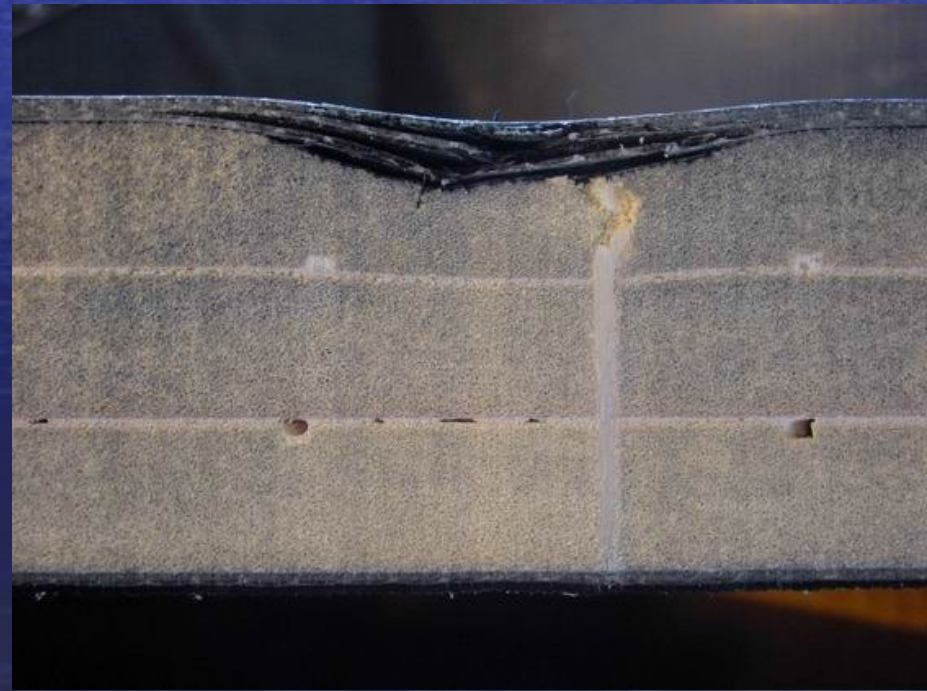
# Sandwich structures

- lighter, because they are stiffer
- cost-effective
- can be more damage tolerant

## Single Skin Laminate- Blunt Projectile Damage



## Sandwich Laminate- Blunt Projectile Damage



# Application of the progress in planing boats design

## Optimization of planing hullforms

- Resistance
- Longitudinal and lateral stability

## Experience with composite hull structures

- Design
- Usage





Amphibious aircraft design is multidisciplinary by nature – there are contradicting requirements for aerodynamics, structural performance and hydrodynamic properties :

### Planing

- Stable Take-off – low drag and spray, longitudinal stability – porpoising
- Hull loads during take-off and landing

### Displacement regime

- Seaworthiness – hull volume
- Lateral stability



# Traditional design of seaplanes

- Use of semi-empirical equations based on statistical data
- Data obtained from model scale tests
- Experience from former projects
- Sequential determination of design parameters

To explore new designs physics based models should be introduced

# Challenges for the high-fidelity CAD based analysis methods (Navier-Stokes fluid flow and FEM structural analyses)

- High complexity of the flow
- Very high computational cost
- Numerical noise due to discretization
- Impossible to explore large design spaces

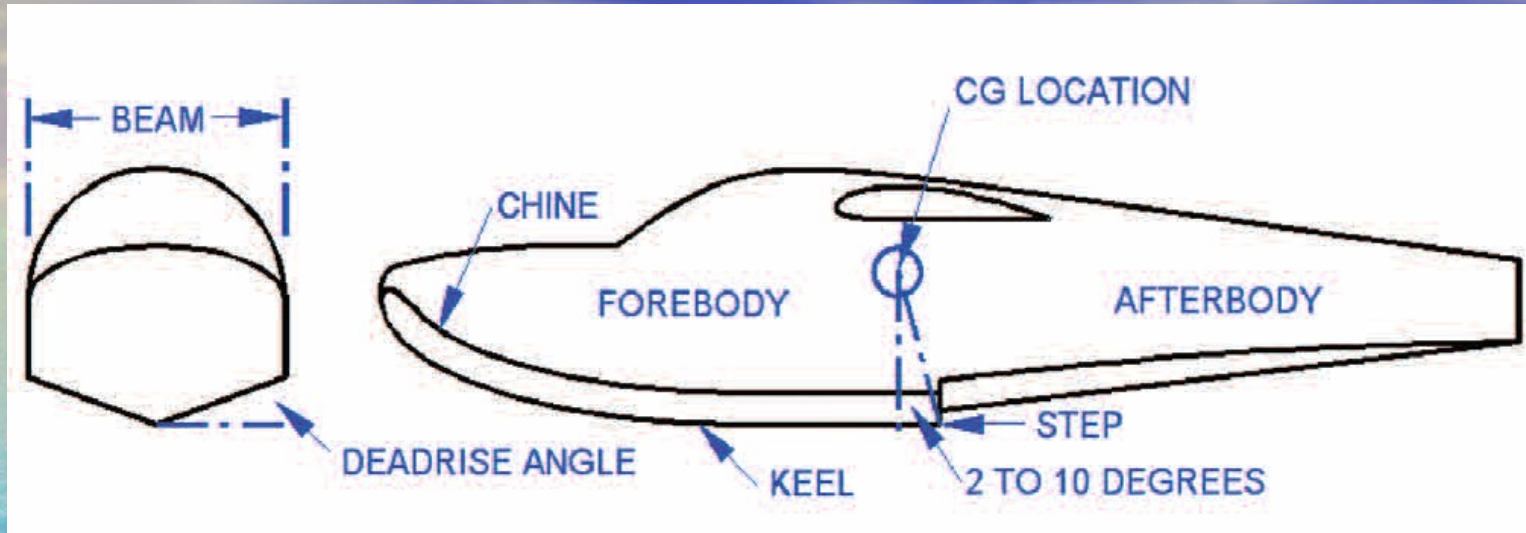
The solution:

Use metamodels (models of models) for  
MDO

# Benefits of metamodels:

- Merging of data from simulation and experimental analysis
- Filtering of numerical noise and experimental errors
- Low computational cost – rapid exploration of the design space
- Possible to use gradient-based optimization methods
- Visualization of the dependencies

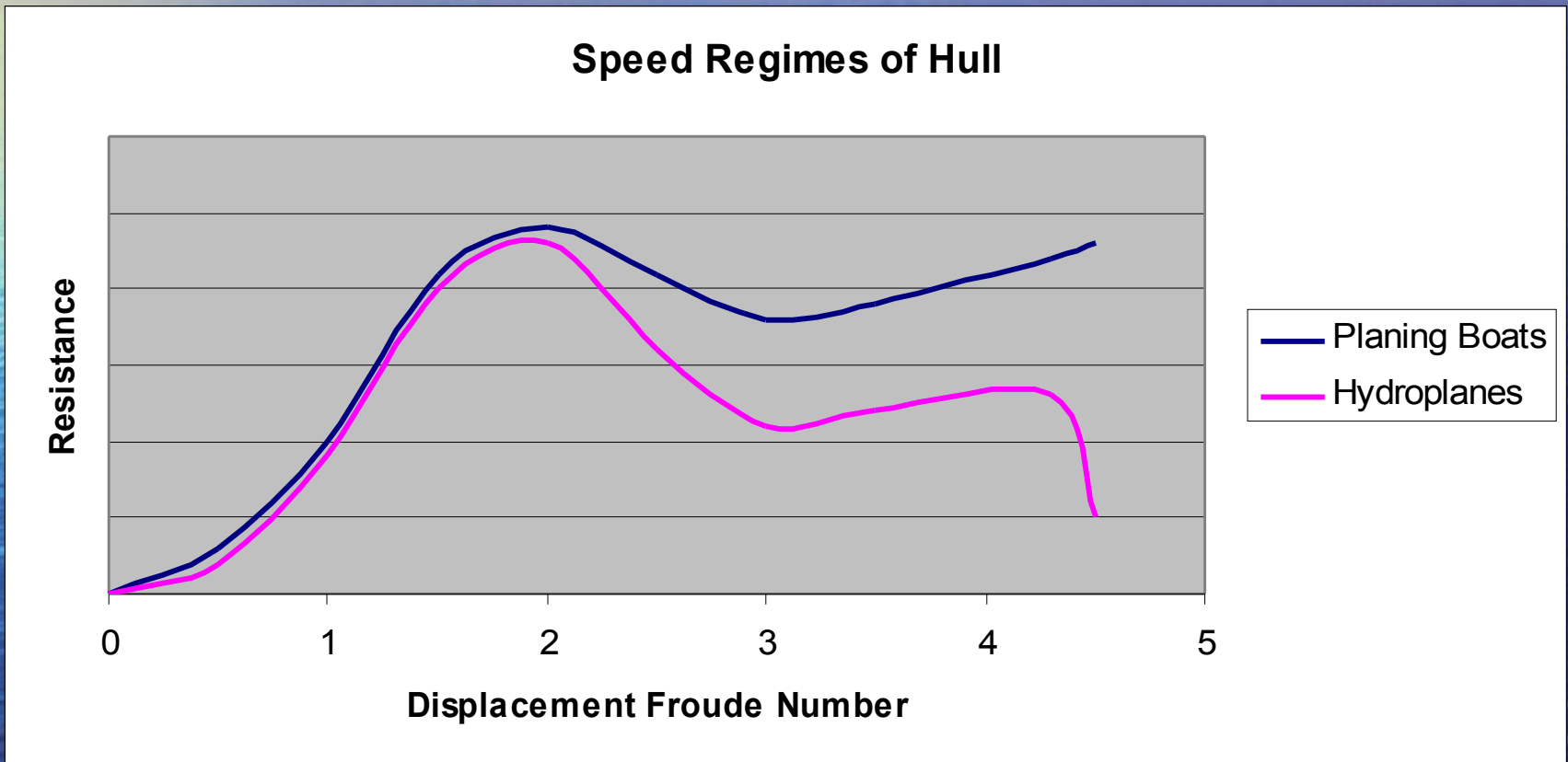
# Flying boat hull definitions



- Beam load coefficient  $C_{\Delta} = \frac{\Delta}{\rho b^3}$

- Displacement Froude Number  $F_{\Delta} = \frac{V}{\sqrt{g^3 \sqrt{\Delta}}}$

# Comparison of the hull resistance of planing boats and hydroplanes



Flying boat design is determined by the take-off condition

Most important parameter - beam

- Classic approach – empirical

Munro[2]       $b = 0,132\sqrt[3]{\Delta}$

$\Delta$  – weight [kg]

b- beam [m]

# Application of planing boats data: Diehl[1]

Determination of beam from the  
hydrodynamic lift coefficient

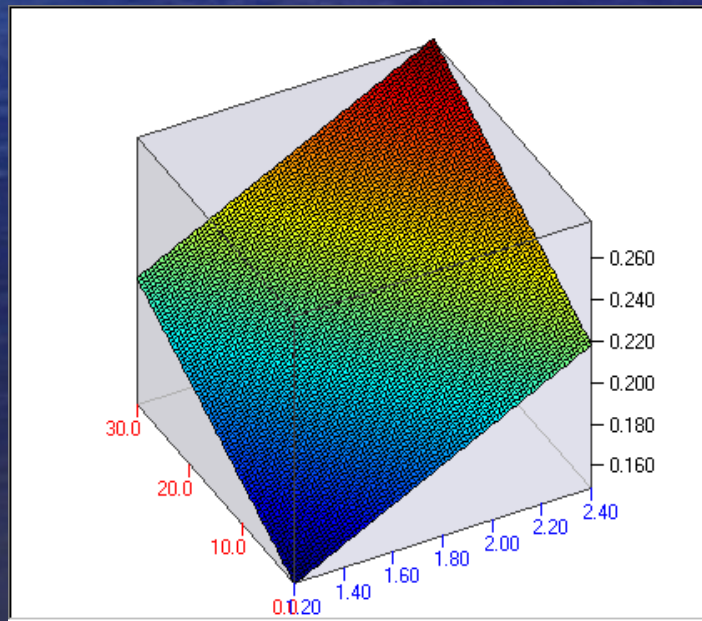
$$C_{hl} = \frac{\Delta}{\frac{\rho V^2}{2} b^2}$$

Beam

$$b = K \sqrt{S}$$

$K = K(\beta, C_{lmax})$

$S$  – wing surface





# Determination of beam for lateral stability in planing

$$b_{\min}(\beta, \Delta)[\text{m}] = 0,5 + 0,0004 \Delta[\text{kg}] - 0,55 \beta[\text{rad}]$$

Longitudinal stability in planing

Forebody length/beam > 3

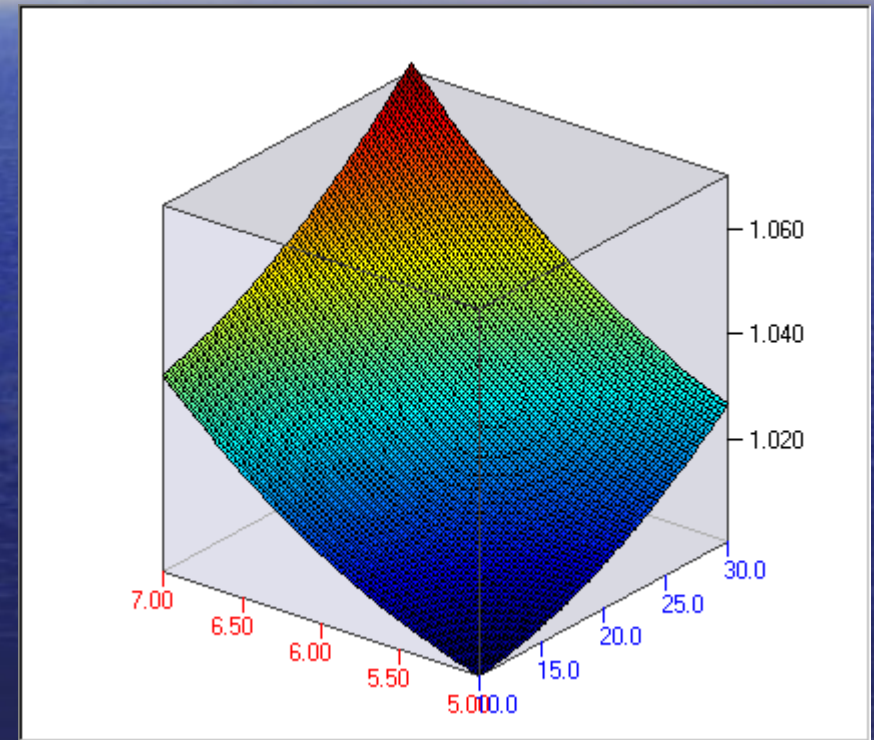
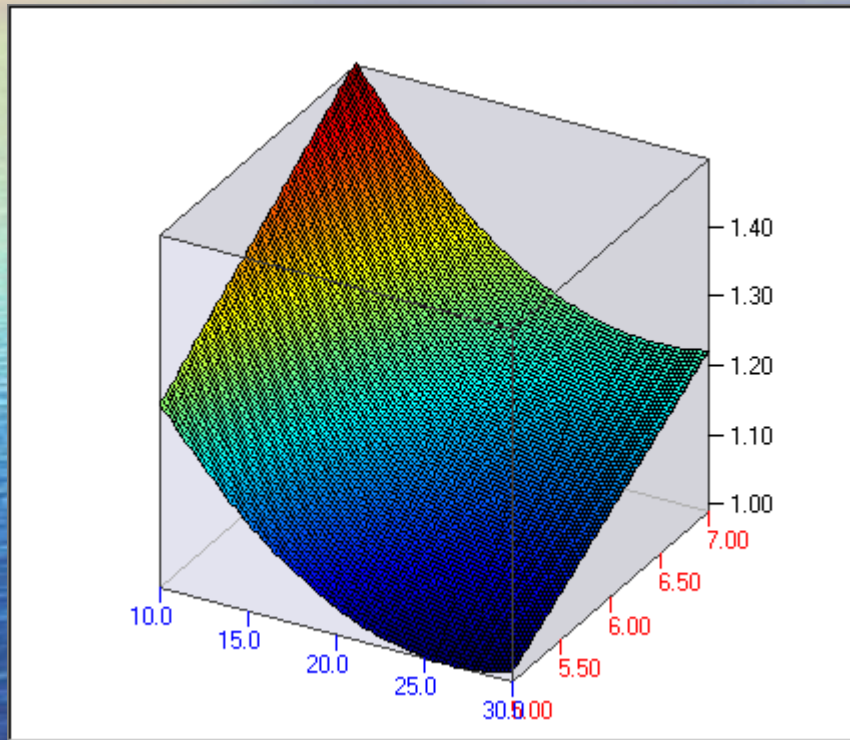
Seaworthiness requirements

Hull volume > 3 \* displacement

# MDO methodology

- Create physics based metamodels for the drag and weight of a seaplane hull as functions of length to beam ratio and deadrise angle
- Determine the constraints from the hull volume requirements and the necessary forebody length
- Calculate the design pressures(CS-23)
- Build a Pareto front and select the design parameters according to mission and seaworthiness requirements.

# Response surfaces

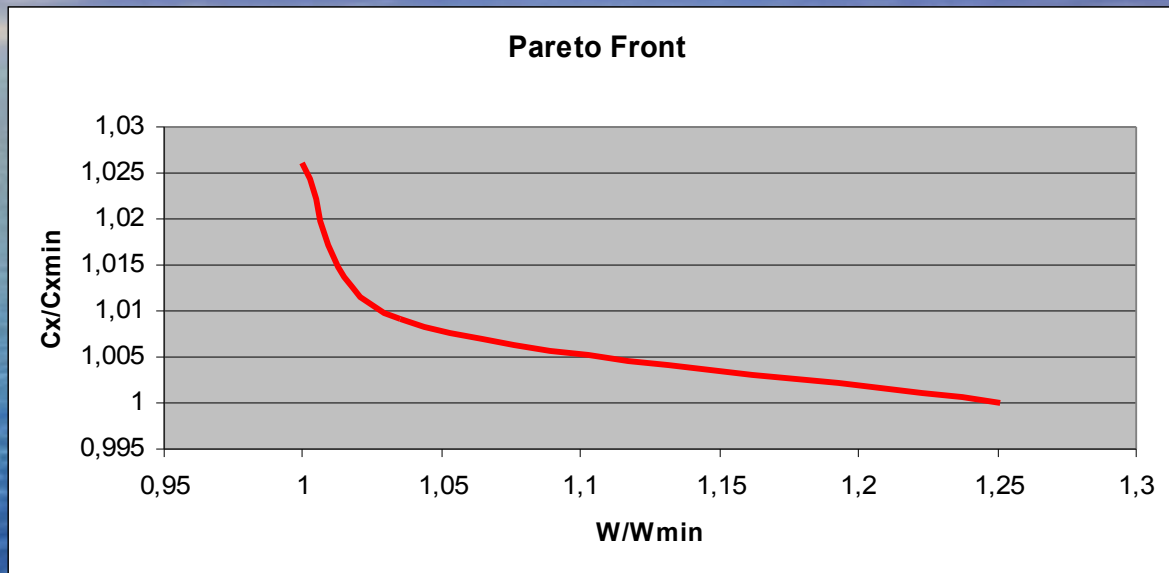


- Weight / min weight ( $L/b, \beta^\circ$ )
- Constant volume of hull

- $C_x / C_{xmin}$  ( $L/b, \beta^\circ$ )

# Pareto front Drag-Weight tradeoff

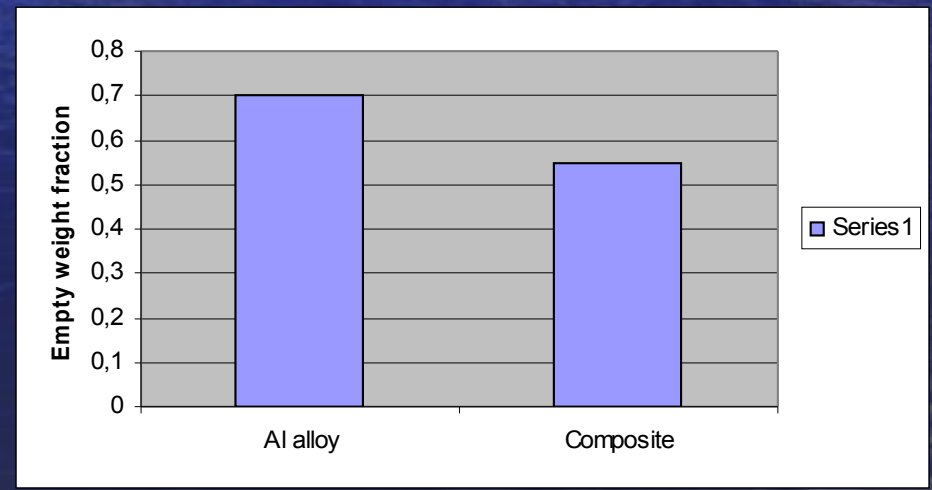
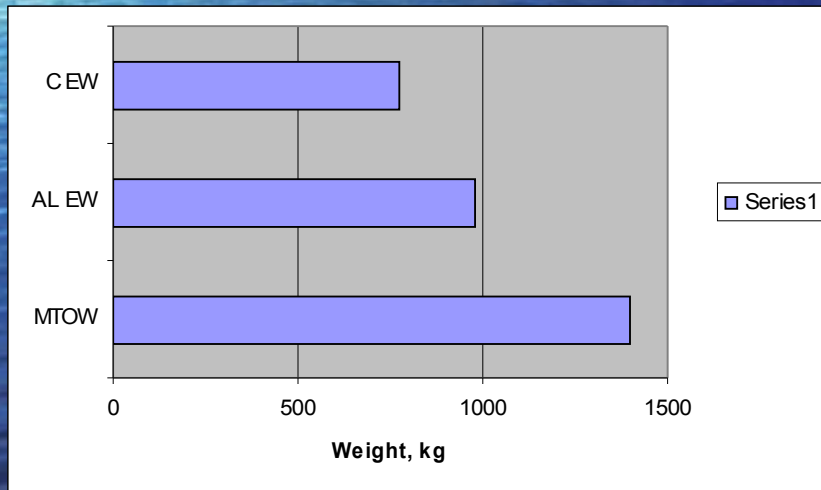
Drag



Weight

# Design Study – Composite amphibious aircraft investigation

The benefits from replacing the Al alloy structure with CFRP sandwich one and optimizing the geometry of the planing hull



# Future Work

- Improve the metamodels with application of kriging or radial basis functions

# References

1. Diehl, W. – The application of basic data on planing surfaces to the design of flying-boat hulls, NACA rep No 696, 1940
2. Munro, W. – Проектирование и расчет гидросамолетов – Москва 1935